

FLAME SPREADING OVER A ROD FUEL IN A QUIESCENT ENVIRONMENT

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ABSTRACT

A theoretical model for flame spread over a rod fuel in a quiescent environment is presented that can be used to predict the structure and spreading rate of flame. The model includes steady-state, two-dimensional momentum, energy and species equations in gas phase and continuity and energy equations in solid phase. In a reference frame attached to the flame front, the flame spreading rate becomes an eigenvalue for this problem. The relationships between flame spreading rate and diameters of fuel rod and ambient oxygen concentration are presented. It is concluded that the flame spreading rate decreases as the diameter of fuel rod is increased whereas is increased as ambient oxygen concentration is increased.

Key Words: *Quiescent environment, Flame spreading, Rod fuel*

NOMENCLATURE

A_s'	Pre-exponential factor for solid pyrolysis	m_f	Fuel mass fraction
B_F'	Frequency factor	m_{OX}	Oxygen mass fraction
C_p'	Gas-phase specific heat	$m_{OX\infty}$	Ambient oxygen mass fraction
C_{ps}'	Specific heat for solid fuel	\dot{m}_s''	Burning rate on solid fuel
D_i'	Dimensional species diffusivity	\dot{M}_s''	Dimensionless burning rate on solid fuel
D_r'	Reference species diffusivity	p'	Pressure
E'	Activation energy for gas phase	p_∞'	Ambient pressure
E	Dimensionless activation energy in gas phase	p	Dimensionless pressure
E_s'	Activation energy for fuel pyrolysis	Pr	Reference Prandtl number
E_s	Dimensionless activation energy of pyrolysis	q_{in}	Heat flux from gas phase to solid phase
Le	Lewis number	r'	Co-ordinate vertical to fuel surface
L_r'	Reference thermal length	r	Dimensionless co-ordinate vertical to fuel surface
L_v'	Latent heat of vaporization	R	Universal gas constant
L_v	Dimensionless latent heat of vaporization		

r_{\max}	Maximum r in computational domain
r_0'	Radius of solid rod
r_0	Dimensionless radius of solid rod
Re	Reynolds number
Sc	Conduction / convective parameter
s	Stoichiometric oxidizer / fuel mass ratio
S_ϕ	Source term for equation for property ϕ
T'	Gas temperature
T	Dimensionless gas temperature
T_f'	Adiabatic stoichiometric flame temperature or flame temperature at the foot of flame
T_∞	Ambient temperature
T_r	Reference temperature
T_s'	Solid temperature
T_s	Dimensionless solid temperature

Greek Symbols

α_r'	Thermal diffusivity of the gas
α_s'	Thermal diffusivity of the solid
α_s	Dimensionless thermal diffusivity of the solid
Γ_ϕ	Effective exchange coefficient for the property ϕ
$\Delta H_c'$	Heat of combustion per unit mass of fuel
ΔH_c	Dimensionless heat of combustion
λ'	Thermal conductivity of gas mixture
λ_r'	Reference thermal conductivity
λ_s'	Thermal conductivity of the solid
γ	Temperature ratio, T_r'/T_∞
μ'	Dynamic viscosity
μ	Dimensionless dynamic viscosity

Subscripts

f	Fuel, flame
g	Gas
ox	Oxygen
r	Reference state (gas phase)
s	Solid

Superscripts

'	Dimensional quantity
•	Per unit time

u'	Velocity parallel to the fuel surface
u	Dimensionless velocity parallel to the fuel surface
v'	Velocity vertical to the fuel surface
v	Dimensionless velocity vertical to the fuel surface
V_f'	Flame spreading rate
V_f	Dimensionless flame spreading rate
V_{ref}	Reference velocity
v_w	Dimensionless interfacial velocity
x'	Co-ordinate parallel to the fuel surface
x	Dimensionless co-ordinate parallel to the fuel surface
x_{bo}	Dimensionless burnout location
x_{\max}	Maximum x in computational domain

μ_r'	Reference dynamic viscosity
ν'	Kinematic viscosity of the gas
ρ'	Gas mixture density
ρ	Dimensionless gas mixture density
ρ'_{bo}	Burnout density of the solid
ρ_{bo}	Dimensionless burnout density
ρ_r'	Reference gas mixture density
ρ_s'	Solid fuel density
ρ_s	Dimensionless solid density
$\rho'_{s\infty}$	Solid density in ambient condition
ρ_w	Gas density at interface
ω_f	Dimensionless reaction rate
$\dot{\omega}_f$	Reaction rate for fuel

w	Solid surface
∞	Ambient condition
max	Maximum
min	Minimum

"	Per unit area
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1 INTRODUCTION

The phenomenon of the propagation of flames over the surface of a combustible material has received considerable attention during the last decade since it relates to the prevention and control of fires and to the fundamental questions in combustion science. The spread of flame results from the complex interaction of transport processes in the gas and condensed phases, the vaporization of the fuel and the chemical reaction of the fuel vapors with the gaseous oxidizer.

Many investigators have studied the problem of flame spread over a thermally thin solid fuel. De Ris^[1] developed the first significant model for spread over thermally thin fuels in an opposing flow that resulted in a simple predictive formula. The paper by Frey and Tien^[2] was the first successful analysis to present the flame spreading rate as a function of opposed flow velocity by numerical methods. They point out the effect of finite rate chemical kinetic on the flame front although the Oseen approximation was retained. Bhattacharjee *et al.*^[3] present a theoretical model that can be used to predict the structure and rate of spread of an attached diffusion flame moving over a thermally thin pyrolyzing combustible placed in a gravity-free, quiescent, oxidizing environment.

However, for the problem of flame spread over a rod fuel, few papers can be seen and more work need to be done. Sibulkin and Lee^[4] measured the flame spreading velocity using PMMA cylinders as fuel rods. Weber and de Mestre^[5] measured the flame spread rate on single ponderosa needles. So far there are few papers on the theoretical study of flame spreading on rod fuel.

A theoretical model for flame spread over a rod fuel in a quiescent environment is presented here and numerical solutions to the model are given. By using the model, effects of rod radius and ambient oxygen concentration on flame spreading rate are studied.

2 MATHEMATICAL MODEL

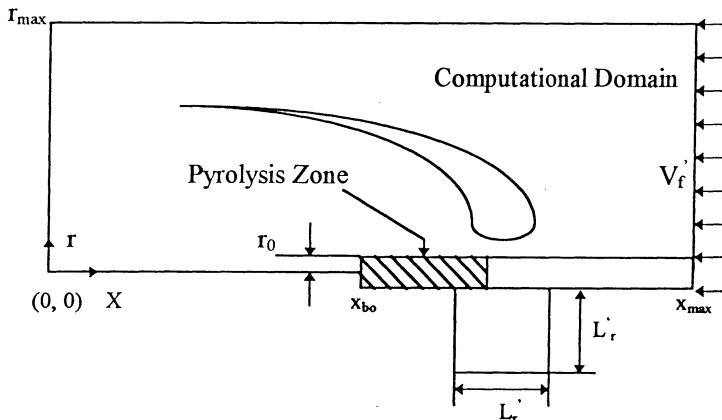
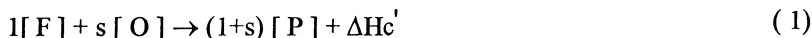


Figure 1 Schematic of the flame spread configuration and computational domain

The specific problem to be studied in this paper is steady diffusion flame spread over a solid rod fuel in a quiescent environment. In a reference frame attached to the flame front, as shown in Figure 1, the flame becomes stationary and the oxidizer environment as well as the solid fuel bed are moving into the flame with velocity V_f' . The flame spread rate is now an eigenvalue and is not known *a priori*.

The model consists of the elliptic, partial differential equations for the conservation of energy, species, mass and momentum in the gas phase and the ordinary differential equations for the conservation of mass and energy in the solid phase. Before introducing the governing equations for this combustion model, several assumptions are made. They are: (1) The flow field is quasi-steady, laminar and cylindrical axis-symmetrical, (2) Radiative heat transfer and body forces are neglected, (3) The fluid is an ideal gas mixture. (4) The gas phase chemistry is described by a one-step overall chemical reaction, which is :



where s is stoichiometric oxidizer / fuel mass ratio and $\Delta H_c'$ is the heat of combustion per unit mass of fuel. The corresponding chemical kinetics is expressed as

$$\dot{\omega}_f = B_F' \rho'^2 m_f m_{OX} \exp\left(-\frac{E'}{RT'}\right) \quad (2)$$

where $\dot{\omega}_f$ is the fuel consumption rate. (5) The solid rod is thermally thin, it means the temperature distribution across it is nearly uniform and the property variations in the r -direction can be assumed negligible. (6) An Arrhenius-typed pyrolysis law is used to describe the fuel gasification:

$$\dot{m}_s = A_s' (\rho_s' - \rho_{bo}') \exp\left(-\frac{E_s'}{RT_s'}\right) \quad (3)$$

(7) Radius of solid rod keeps constant and only density varies during the fuel gasification.

With these assumptions and from a reference frame at the flame front, we get the gas-phase formulation and the solid-phase formulation. The gas-phase formulation is consisted of six quasi-steady partial differential equations and a one-step overall reaction. The solid-phase formulation includes two ordinary differential equations and a Arrhenius kinetic.

From Bhattacharjee *et al.*^[3], conduction through the solid is unimportant, the thermal length in the gas phase, $L_T' = \alpha_T' / V_{ref}$, is used as the length scale in both phase. Reference property such as α_T' , etc. are evaluated at a reference temperature T_r' , which is the average of the adiabatic, stoichiometric flame temperature without dissociation and the ambient temperature.

The dimensionless variables and parameters are defined as:

$$\begin{aligned} L_r' &= \frac{\alpha_r'}{V_{ref}} \quad x = \frac{x'}{L_r'} \quad r = \frac{r'}{L_r'} \quad u = \frac{u'}{V_{ref}} \quad v = \frac{v'}{V_{ref}} \quad \rho = \frac{\rho'}{\rho_r'} \quad p = \frac{p' - p_\infty'}{\rho_r' V_{ref}^2} \quad \mu = \frac{\mu'}{\mu_r'} \quad T = \frac{T'}{T_\infty} \\ E &= \frac{E'}{RT_\infty} \quad \Delta H_c = \frac{\Delta H_c'}{C_p' T_\infty} \quad Re = \frac{V_{ref} L_r' \rho_r'}{\mu_r'} \quad Pr = \frac{v'}{\alpha'} \quad Le = \frac{\alpha'}{D'} \quad Da = \frac{B_F' \rho_r' L_r'}{V_{ref}} \\ V_f &= \frac{V_f'}{V_{ref}} \quad T_s = \frac{T_s'}{T_\infty} \quad E_s = \frac{E_s'}{RT_\infty} \quad \rho_s = \frac{\rho_s'}{\rho_{s\infty}'} \quad A_s = \frac{A_s' L_r' (\rho_{s\infty}' - \rho_{bo}')}{V_{ref} \rho_{s\infty}'} \quad \dot{M}_s = \frac{\dot{m}_s' L_r'}{V_{ref} \rho_{s\infty}'} \\ \alpha_s &= \frac{\alpha_s'}{L_r' V_{ref}} \quad L_v = \frac{L_v'}{C_{ps}' T_\infty} \quad Sc = \frac{\lambda_r'}{\rho_{s\infty}' C_{ps}' L_r' V_{ref}} \quad \dot{\omega}_f = \frac{\dot{\omega}_f' L_r'}{V_{ref} \rho_r'} \end{aligned} \quad (4)$$

where subscript " r " represents quantities in the reference state at temperature T and " s " represents properties of the solid phase.

2.1 Governing equations

The nondimensional governing equations and boundary conditions for both phases are as following.

2.1.1 Gas-phase governing equations

All the conservation equations in the gas-phase can be written in the general form:

$$\frac{\partial}{\partial x}(\rho u \varphi) + \frac{\partial}{r \partial r}(r \rho v \varphi) = \frac{\partial}{\partial x}(\Gamma_{\varphi} \frac{\partial \varphi}{\partial x}) + \frac{\partial}{r \partial r}(r \Gamma_{\varphi} \frac{\partial \varphi}{\partial r}) + S_{\varphi} \quad (5)$$

where φ , Γ_{φ} and S_{φ} for the different equations are shown in Table 1. In Table 1, the dimensionless reaction rate for the fuel vapor is defined as $\dot{\omega}_f = \text{Dap}^2 m_f m_{\text{OX}} \exp(-E/T)$.

Table 1 — Values of φ , Γ_{φ} and S_{φ}

	φ	Γ_{φ}	S_{φ}
Continuity	1	0	0
x-Momentum	u	$\frac{\mu}{\text{Re}}$	$-\frac{\partial p}{\partial x} + \frac{\partial}{\partial x}(\frac{\mu}{\text{Re}} \frac{\partial u}{\partial x}) + \frac{\partial}{r \partial r}(\frac{\mu}{\text{Re}} r \frac{\partial v}{\partial r})$
r-Momentum	v	$\frac{\mu}{\text{Re}}$	$-\frac{\partial p}{\partial r} + \frac{\partial}{\partial x}(\frac{\mu}{\text{Re}} \frac{\partial u}{\partial r}) + \frac{\partial}{r \partial r}(\frac{\mu}{\text{Re}} r \frac{\partial v}{\partial r}) - \frac{2\mu}{\text{Re}} \frac{v}{r^2}$
Fuel	m_f	$\frac{\mu}{\text{Re Pr Le}}$	$-\dot{\omega}_f$
Oxygen	m_{OX}	$\frac{\mu}{\text{Re Pr Le}}$	$-s \dot{\omega}_f$
Energy	T	$\frac{\mu}{\text{Re Pr}}$	$\Delta H_c \dot{\omega}_f$

The six equations in Table 1 along with the equation of state $\rho T = \gamma$ are sufficient number for determining the seven unknowns u, v, p, T, ρ , m_f and m_{OX} . Viscosity μ is determined by

$$\mu = \frac{T}{\gamma} \quad (6)$$

2.1.2. Solid-phase governing equations

Continuity equation:

$$\dot{M}_s'' = V_f \frac{d\rho_s}{dx} = A_s \frac{\rho_s - \rho_{bo}}{1 - \rho_{bo}} \exp(-\frac{E_s}{T_s}) \quad (7)$$

Energy equation:

$$\rho_s V_f \frac{dT_s}{dx} + \alpha_s \frac{d^2 T_s}{dx^2} = \dot{M}_s'' L_v - \frac{2}{r_0} \mu Sc \frac{\partial T}{\partial r} \Big|_w \quad (8)$$

2.2 Boundary Conditions

2.2.1 Gas-phase boundary conditions

At $x = x_{\text{max}}$:

$$u = -V_f' / V_{\text{ref}} \quad v = 0 \quad m_f = 0 \quad m_{\text{OX}} = m_{\text{OX}\infty} \quad T = 1 \quad (9)$$

At $x = 0$:

$$\frac{\partial u}{\partial x} = \frac{\partial v}{\partial x} = \frac{\partial m_f}{\partial x} = \frac{\partial m_{ox}}{\partial x} = \frac{\partial T}{\partial x} = 0 \quad (10)$$

At $r = r_{\max}$:

$$u = -V_f^l/V_{\text{ref}} \quad \frac{\partial (r\rho v)}{\partial r} = 0 \quad m_f = 0 \quad m_{ox} = m_{ox\infty} \quad T = 1 \quad (11)$$

At $r = r_0$:

for $0 < x < x_{b0}$

$$\frac{\partial u}{\partial r} = V = \frac{\partial m_f}{\partial r} = \frac{\partial m_{ox}}{\partial r} = \frac{\partial T}{\partial r} = 0 \quad (12)$$

for $x_{b0} < x < x_{\max}$

$$u = -V_f^l/V_{\text{ref}} \quad v = v_w \quad T = T_s$$

$$\rho_w V_w \varphi - \frac{1}{\text{Re Pr Le}} \frac{\partial \varphi}{\partial r} \Big|_W = \begin{cases} \rho_w V_w & (\varphi = m_f) \\ 0 & (\varphi = m_{ox}) \end{cases} \quad (13)$$

$$\rho_w V_w = I \dot{M}_s^*, \quad I = 0.5 V_{\text{ref}} \rho_{\infty} r_0 C_p / \lambda_r, \quad (14)$$

V_f^l, v_w and T_s are obtained from solution to the solid-phase problem

2.2.2 Solid-phase boundary conditions:

At $x = x_{\max}$:

$$\rho_s = 1 \quad T_s = 1 \quad (15)$$

At $x = x_{b0}$:

$$\rho_s = \rho_{b0} \quad \frac{dT_s}{dx} = 0 \quad (16)$$

where ρ_{b0} is introduced to determine the three unknowns, T_s , ρ_s and V_f . The density at burnout point is set to a specified value, $\rho_{b0} = 6\%$. Heat flux at the interface, $(2/r_0)S_c \mu \partial T / \partial r \Big|_w$, is obtained from the gas phase solution.

From the equations above-mentioned, we can see that gas-phase equations and solid-phase equations are coupled each other through the boundary conditions at the interface. The solution of gas-phase problem provides heat flux at the interface to the solid-phase equations and the solution to the solid-phase equations provides the boundary conditions for the gas-phase equations. At the same time, V_{ref} is not a constant during calculation. It is taken to be V_f^l . However, V_f^l is contained in several dimensionless parameters, such as A_s , α_s , Sc etc.. Since V_f^l is not known *a priori*, an iteration scheme is needed to solve the problem in both phases. Newton Method^[7] is used to solve the nonlinear ordinary differential equations in the solid phase. The gas-phase equations are solved using the SIMPLE algorithm^[6]. The computer code was developed and modified by the author.

The computational domain in the gas phase was 20×8 . For all cases, the burnout location, x_{b0} , was fixed at $x = 6$, and the flame tip location was allowed to change during the iteration process.

3 RESULTS AND DISCUSSION

To show the effects of diameters of fuel rods and oxygen concentration on flame spreading velocity, 9 cases are calculated. They are shown in Table 2. For all cases, properties of gas and solid are the same. All of the properties are summarized in Table 3. The gas-phase thermodynamic properties are evaluated at reference temperature T_r' , i.e., $T_r' = (T_f' + T_\infty)/2$, where T_f' is the adiabatic stoichiometric flame temperature.

Table 2 Radius of fuel rod and ambient oxygen concentration for every case

case No.	r_0 (mm)	$m_{O_2\infty}$	case No.	r_0 (mm)	$m_{O_2\infty}$	case No.	r_0 (mm)	$m_{O_2\infty}$
1	1	0.233	4	6	0.233	7	2	0.4
2	2	0.233	5	8	0.233	8	2	0.5
3	4	0.233	6	2	0.3	9	2	0.6

Table 3 Gas and solid property values

Symbol	Units	Value	Reference
ρ_r'	kg/m ³	0.227	Natl. Bur. Stan. (1955)
μ_r'	kg/m·s	5.55×10^{-5}	Natl. Bur. Stan. (1955)
λ_r'	W/m·K	9.8×10^{-2}	Natl. Bur. Stan. (1955)
C_p	J/kg·K	1.239×10^3	Natl. Bur. Stan. (1955)
α_r'	m ² /sec	3.478×10^{-4}	Natl. Bur. Stan. (1955)
R	J/mole·K	8.314	Natl. Bur. Stan. (1955)
s	—	1.185	Altenkirch <i>et al.</i> (1980)
T_f'	K	2822	Altenkirch <i>et al.</i> (1980)
T_∞	K	298	Altenkirch <i>et al.</i> (1980)
ΔH_c	J/kg	1.674×10^7	Altenkirch <i>et al.</i> (1980)
E	J/mole	7.428×10^4	Duh and Chen (1990)
B_F	m ³ /kg·s	3.125×10^7	Bhattacharjee (1990)
A_s	1/sec	1.0×10^{10}	Frey and T'ien (1977)
E_s	J/mole	1.255×10^5	Frey and T'ien (1977)
λ_s	W/m·K	1.255×10^{-1}	Frey and T'ien (1977)
C_{ps}	J/kg·K	1.26×10^3	Altenkirch <i>et al.</i> (1980)
L_v	J/kg	7.53×10^5	Altenkirch <i>et al.</i> (1980)
$\rho_{s\infty}$	kg/m ³	750	Altenkirch <i>et al.</i> (1980)
α_s	m ² /sec	1.328×10^{-7}	Duh and Chen (1990)

3.1 Flame Spreading Rate

Flame spreading rate (V_f) at different diameters of fuel rod is shown in Figure 2. It can be seen that the flame spreading rate decreases as the diameter of fuel rod increases.

Figure 3 shows the relationship between flame spreading rate and ambient oxygen concentration. It can be seen that flame spreading rate increases as the ambient oxygen concentration increases.

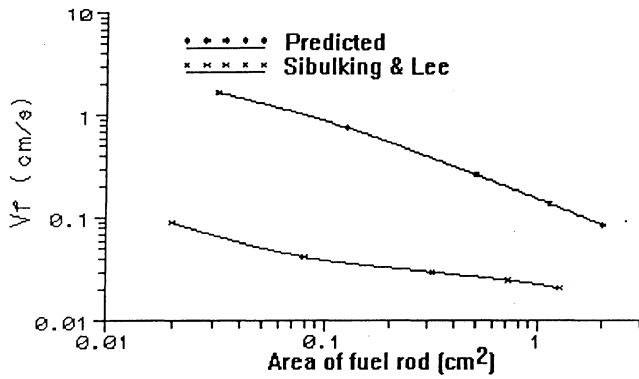


Figure 2 Flame spreading rate versus area of fuel rod

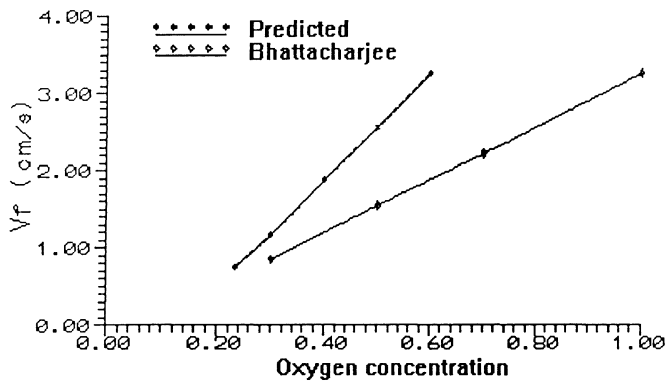


Figure 3 Flame spreading rate versus ambient oxygen concentration

3.2 Flame Structure

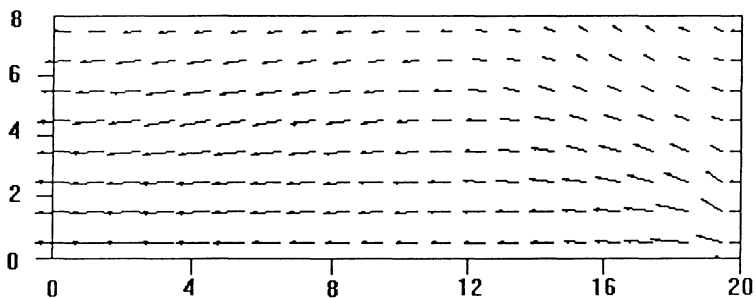


Figure 4 Velocity vectors distributions (frame attached to flame front)

Figure 4 shows the velocity vectors in gas phase of case No. 2. At reference frame attached to flame front, the flow pattern is similar to that in an opposed flow. The difference is that opposed flow velocity is the same as flame spreading rate in this case.

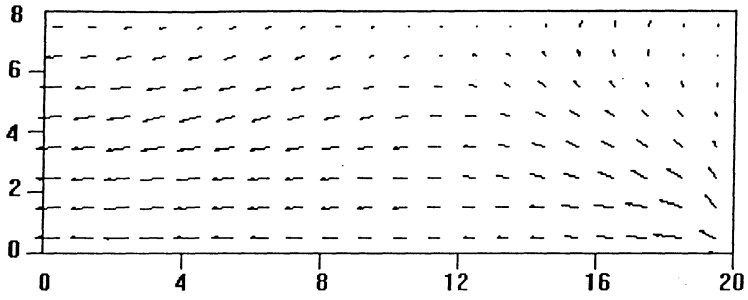


Figure 5 Velocity vectors distributions (at laboratory coordinate)

The flow pattern in gas phase at laboratory coordinate is shown in Figure 5.

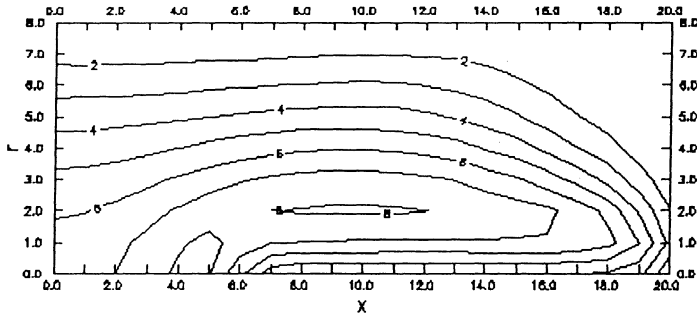


Figure 6 Isotherm distributions (dimensionless temperature)

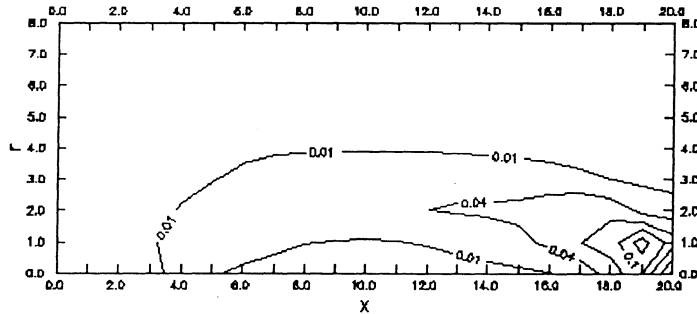


Figure 7 Dimensionless fuel reactivity rate distributions

Figure 6 shows the isotherm distributions in gas phase of case No. 2. It is found that the temperature gradient at preheat zone is great, which results a large heat flux at interface between gas phase and solid phase. (See Figure 8)

The gaseous fuel reactivity distribution, defined as $\omega_f = Da\rho^2 m_f \exp(-E/T)$, is shown in Figure 7. It can be seen combustion takes place mainly in the pyrolysis region. Since the fuel vapors are not completely consumed in the pyrolysis region, the flame extends into the downstream wake region.

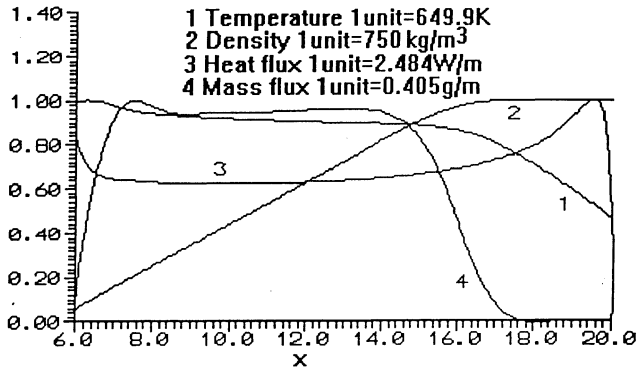


Figure 8 Solid phase response

The response of solid fuel is shown in Figure 8. It includes the distributions of fuel temperature, density, heat flux and the mass flux from the gas phase. It can be seen that the temperature of fuel rod increases rapidly as it reaches the flame front, and then increases gradually. The density of fuel rod keeps constant first, then it decreases almost linearly.

An interesting phenomenon is that the maximum q_{in} takes place in preheat region. Since the gaseous fuel reactivity, ω_f , is large, the temperature difference between gas phase and solid phase get its maximum, which results the maximum of q_{in} .

3.3 Comparisons with Experiment and Other Related Studies

The most relevant experimental study for comparison to this work is that of Sibulkin and Lee^[4]. In their paper, flame spreading velocity measurements were made using PMMA cylinders as fuel rods. The comparison between predicted results and that of Sibulkin and Lee^[4] is shown in Figure 2. It can be seen that predicted results are the same as that of Sibulkin and Lee qualitatively. The difference between them is that predicted results are based on filter paper.

The most relevant numerical study is that of Bhattacharjee *et al*^[3]. In their paper, flame spreading over a thermally-thin fuel bed is calculated. Comparisons between their results and ours are shown in Figure 3. It can be seen that relationship between flame spreading rate and ambient oxygen concentration given by Bhattacharjee *et al* is almost the same as that in this paper. The difference between them is that flame spreads faster over rod fuel than over fuel bed at same ambient oxygen concentration.

4 CONCLUSIONS

A mathematical model for flame spread over a cylinder fuel in a quiescent environment is developed and solved numerically to predict the structure and spreading rate of flame. In a reference frame attached to the flame, a quasi-steady state in the gas phase with respect to the solid phase can be established and the solid-phase equations become steady and the fuel is fed into the flame with velocity V_f . The gas-phase equations are coupled to the solid-phases ones through the spreading rate, which appears as an eigenvalue in both phases, and the interfacial boundary conditions. The mathematical

system consists of the two-dimensional momentum, energy, species and continuity equations in the gas phase. A one-step overall chemical reaction with second-order Arrhenius kinetic is also included. In the solid phase, the energy balance equation and mass balance equation are included. The solid-phase mass balance includes an Arrhenius expression for fuel pyrolysis.

The mathematical model is solved numerically. SIMPLE^[6] algorithm is adopted to solve the partial differential equations in the gas phase and Newton Method^[7] with an under-relaxation is used to solve the nonlinear ordinary differential equations in the solid phase. Iteration scheme is made between gas phase and solid phase.

According to the simulation results, the flame spreading rate decreases as the diameter of fuel rod is increased. The flame spreading rate decreases as the ambient oxygen concentration decreases.

ACKNOWLEDGMENTS

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