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Research Programme Objective

DEPARTMENT OF SCIENTIFIC AND INDUSTRIAL RESEARCH AND FURE OFFICES' COMMITTEE JOINT FIRE RELEASE OF DANIZATION

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> > FOAM FOR ALACRAFT CRASH FIRES (1)

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D. Hich, R. J. French and P. Nash

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Fire Research Station, Boreham Wood, Herts.

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FOAM FOR ALHOHAFT CHASH FIRES (1)

by

D. Hird, R. J. French and P. Nash

1. Introduction

Aircreft crash fires differ in many ways from other flammable liquid fires, such as those in petrol storage tarks. The time element is usually critical in that people must be rescued at the earliest possible moment, and the account therefore lies in the rapid reduction of the intensity of the fire rather than on its extinction. It is unlikely that there will be any appreciable area of free petrol surface, and in order to obtain rapid coverage of the aircraft orash and the surrounding terrain, the form must be applied directly to the burning areas.

Foam may be applied either as a jet or a spray, and opinions of operational officers differ on the relative merits of the two methods. While many tests have been made with sprayed foam on large spill fires, there is little information on the application of foam as a jet to this type of fire. An initial programme of tests was therefore planned to see whether there was any marked advantage in using one method of application in preference to the other, and at the same time the effects of varying the expansion and oritical shearing stress of the foam were investigated.

2. Previous work on sprayed foam

Results of large-scale tests by the Engineer Research and Development Laboratories were reported in 1950 (1). A petrol fire 25 ft x 50 ft fas tackled using a number of "fog foam" nozzles. The majority of these nozzles were water spray nozzles of the impinging jet type which, when used with a 6 per cent foam solution, gave a low expansion rapidly-draining foam. One of the nozzles tested was a self-aspirating nozzle similar to a branch pipe with a fan-shaped diffusor which appeared to give a foam of expansion about 6. The results of these tests are shown in Fig. 1.

Experimental work at the Naval Research Laboratories (2) was carried out at one rate of application, 0.083 gal/ft²/min. Preliminary tests on an openended 2 ft diameter drum of petrol were followed by tests on a 400 ft² tray with about $\frac{1}{2}$ in. of petrol an a mud base. The spray was produced from a fixed nozzle with sixteen outlet pipes, the diameters of which were varied to obtain satisfactory performance with the foams of different expansion which were tested The conclusions drawn from these tests were that there was an optimum expansion of 10-12 when foam was applied in a dispersed pattern and that variations in viscosity at a given expansion were of no direct importance.

5. Experimental proocdure

3.1. Simulated aircraft crash fire

The test fire (Fig. 2) was arranged to simulate, in a simple manner, the main features of an aircraft crash fire. A 10 ft by 10 ft bund was constructed and two diagonally opposite quarters were built up with rubble and sand; the other two quarters were left unobstructed and a water level was maintained just below the sand surface. Four 40 gal, drums were placed symmetrically in the bund with their centre lines diagonally across the bund. About 15 gal, of petrol (notor spirit) was used for each test. The fire thus constated partly of free petrol surface and partly of petrol-seaked sand, with a simple representation of an aircraft fuselage. Because of the symmetry of this errangement the fire could be attacked from any side to suit the direction of the prevailing wind. A 30-second proburn was allowed before fire-fighting was commenced. The progress of the fire control was recorded by means of radiometers arranged in the manner used in previous surface application experiments (3). The operator approached was fire from the windward side and moved round the bund to reduce the fire as quickly as possible. The criterion used to determine the efficiency of extinction was to measure the "90 per cent" control time, i.e. the time taken to reduce the radiation from the fire to one-tenth of its initial intensity. The size of the fire at this stage was considered to be such that a suitably clothed trained man could walk through the fire area to make a rescue attempt. No tests were carried out when the wind speed was more than 10 ft per sec.

Twenty-four tests were made, the order being selected to minimize the effect of the operator's learning factor. In half the tests, jet application was used and in the other half the foam was applied as a suray. For each method of application three liquid rates were used at two levels of critical shearing stress for each of two foam expansions. The two expansions were 7 and 14, the liquid rates of application were 0.05, 0.075 and 0.10 gal/ft⁻²/min⁻¹ and the two ranges of critical shearing stress were 300-450 dynes/cm² and 600-850 dynes/cm².

3.2. Free petrol and petrol-snaked sand fires

Following the twenty-four tests described above, a smaller number of tests were made to compare the use of jet and spray application on two of the basic elements of the aircraft crash fire, namely, fire on a free petrol surface and on petrol-soaked ground.

The same 10 ft square bund was used. For the free petrol tests 15 gal of fucl was floated on water, while for the other tests the whole bund was filled with rubble and finished off with a level sand surface. The test procedure was similar to that used for the earlier experiments and foam of expansion 14 and critical shearing stress 450 dynes/cm² was used. For the free petrol fires the jet was always directed into the petrol, no use being made of the bund wall.

4. Experimental results and discussion

used to suit the foar flow rate.

4.1. Simulated aircraft crash fires

The results of the buenty-four tests are shown in Table 1 and Fig. 4 and the results of the statistical analysis are given in Appendix 1. The analysis shows that of the four variables all but the expansion have a significant effect on the time taken to control the fire, the rate of application and critical shearing stress of the form having a greater effect than the method of application. The "most likely" results can be calculated using only the significant effects and these are plotted in Fig. 5.

TABLE I

RESULTS OF TESTS ON 100 SQ. FT MOCK АТНСКАЕТ FIRE

	_	•	. , ? 9/	10 .			
EY PANSTON	CRITICAL SHEARING STARES	METHOD OF APPLICATION	1/10 CONTROL TIME (SECS.) RATES OF APPLICATION: GAL.ft-2min-1				
			0.10	0.075	0.05		
	LOW	JEF	43 •0715	48 • 060	45		
7	yep	SPRAY	47	42	55 • 04 b		
	HIGH 7 ⁰⁰	JET	29 •0UT	56 •070	77		
		SFRAY	46	45 •056	103 . 086		
	T-CHJ	JET	28 •047	34	48.040		
14	2001	SPRAY	37	36 • NUS	48 • 040		
· · · · · · · · · · · · · · · · · · ·	HIGH	JET	42	43 • 054	⁶⁰ . 05 0		
	:	SPRAY	56	44 , oss	⁸⁸ .073		

The effects of the foam properties on the control time were similar for both jet and spray application. In both cases there was no significant effect of expansion and at the lower rates of application the more fluid foams controlled the fire more rapidly than the stiff feams. As the rate of application from the jet increased, the control time decreased as would be expected. However, the shape of the control time/rate curves for spray application, particularly with the low critical shearing stress, show that it tended to take longer to control the fire at the high rate of application than at the medium rate. To interpret this and the difference between the two methods of application it is necessary to consider what happens as the foam passes through the flames.

Water will be lost from the foam by evaporation as it passes through the flames, and this loss will depend on the expansion of the foam, the flake size and the time the foam takes to pass through the flames. These evaporation losses have been estimated in Appendix 2. The way in which they depend on the properties of the foam spray can be seen by making an approximation to equation 3 of Appendix 2. This can be re-written -

fraction of the water in form reaching the petrol surface

$\frac{M}{M_o} = (1 - \frac{B_o t_o E}{r_o})^3$

where $\frac{m}{M}_{O}$ 18 M

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time for which flakes are in the flame. . 8

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initial radius of foam flakes ro =

= length of path through flamo

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constant ۰. B = constant

'expansi on

Now t velocity of foam flakes (v)

It it is assumed that the flakes are travolling with their terminal velocity then for flakes less than about 1 om in diemeter 10. E.S.

 $\frac{M}{M_{b}} = \left[\frac{1 - B'E'^{1/2}}{\gamma_{b}^{2/14}}\right]^{3}$

and for flakes greater than about 1 cm in diameter

$$\frac{M}{M_0} = \left[1 - \frac{B' \cdot E''}{\gamma_0''} \right]^3$$

where B' and B" are constants.

It can be seen that the evaporation losses from the foam in the form of a jet would be insignificant but in a spray would depend on both the expansion and the flake size of the foam. For flakes less than about 1 on diameter the losses are more dependent on the flake size than on the expansion. The percentage of water reaching the petrol surface under the conditions of the tests have been estimated and are shown in Fig. 6.

Because of the importance of the flake size in the foam sprays samples of the sprays were collected on glass plates and photographed. About one hundred flakes of each spray were sized and the mass median flake size estimated; these are shown in Table 2.

Expansion	Critical. shearing stress	Rate of application gal/ft ² /min	MASS MEDIAN flake diameter (cm)		
		0+1	0.8		
	Low	0.075	<u>,</u> 1•3		
7		0.05	æ		
		0,1	0.5		
	High	0.075	1.2		
		0.05	X		
		l 0.1	0.4		
14	Low	0.075	0.6		
		0.05	1.0		
		0.1	0.5		
	High	0.075	0,4		
· · · ·		0.05	0.75		

Table 2. Mass median flake size of foam sprays

The sprays produced with feam of expansion 7 at the lowest rate of application were really a series of small jets and no estimate of the flake sizes were made.

It can be seen from Table 2 that the flake sizes tended to decrease as the rate of application increased and also that the smaller flakes were generally obtained with the high expansion form.

Considering then the results obtained with sprayed form (Fig. 5) the increase in control time as the rate increases from 0.075 to 0.1 gal/ft²/min may well be explained by the reduction in flake size increasing the evaporation losses and reducing the rate of application to the petrol surface. However, one would expect an increased evaporation loss from the high expansion form on two

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counts; firstly, theory suggests that for the same flate size the evaporation losses are greater with the high expansion foam, (Fig. 6), and secondly, at the same rate of application the flake sizes of the high expansion foams were found to be less than those of the low expansion foams. Despite this the total effect of expansion was not significant. This indicates that if the spray consisted of larger flakes, with which the evaporation losses would be negligible, the high expansion foam might control the fire more rapidly than the low expansion foam.

A. C. S. S.

The tests showed, therefore, that when foam was applied to the fire as a jet the control time decreased as the rate of application increased and the fluid foams controlled the fires more rapidly at the two lower rates of application than did the stiffer foams. There was no significant difference between foams of expansion 7 and expansion 14.

When the foam was applied as a spray the fluid foam again controlled the fire more rapidly at the lower rates of application and the effect of expansion was insignificant. However, since the flake sizes of the foam were in the region where considerable losses by evaporation were possible, it would be unwise to draw any firm conclusions on either the effects of expansion or the relative merits of spray and jet application had the spray consisted of flakes large enough for the evaporation losses to be negligible.

One other factor may affect a comparison between the application of foam as a jet and as a spray. A jet of foam breaks up to some extent and by manipulation a dispersed pattern can be achieved over a reasonable area. It may be, however, that there is an effective limit to this, and the advantages of spray would show up more when large areas are involved.

4.2. Free petrol surface and petrol-soaked send

The control time rate of application relationships for these tests are shown in Fig. 7. On the free petrol surface there is little indication of any difference in the two methods of application, although the reservations given above will apply particularly as a high-expansion foam was used. In the tests with petrol-soaked sand the jet took much longer to control the fire than did the spray. The foam was rapidly broken down in contact with the sand when applied as a jet and after extinction little foam remained. This effect may have been largely mechanical.

Conclusions

At this stage of the investigations it is difficult to come to any very definite conclusions as to the influence of the method of applying form to this type of fire. There is some indication that spray application may be better if the flakes are of comparatively large size, so that most of the form reaches the petrol surface.

The most important form property when form is applied as a jet is the critical shearing stress; a fluid form controlling the fire. more rapidly than a stiff one. With this method of application there was no significant difference between forms of expansion 7 and 14.

A fluid form is also an advantage with spray application but in these tests the effects of expansion may may well have been shrouded by the variation of flake size with expansion.

Loklowledgments

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- (4) Heat Transmission. W. H. MoAdams. MoGraw Hill.

	1			<u>AF</u> RESULTS OF S	PENDIX I TATISTICAL ANALYSI	<u>IS</u>	•	1		
Man result:	Standard cm.or	Coeff. of	Mean Values of significant effects							
Statistic	sec.	of observations	variatión %	Linear Effect of R (R ₁)	Curvilinear offect of K (R ₂)	Linear effect of M-(M1)	Linear effect of S-(S ₁)	R2 ^M 1	B ₁ S ₁	
Control Time	50	<u>+</u> 8.2	16.4	- 12,25 ^{*28}	+ 3.25 ^{**}	+ 3.9 [¥]	+ 7.42 ³⁷³⁸	+2 .8 4	-7.15 ⁸⁰⁸	•
Factor S. E. Factor S. E. Factor E. E.	ffect of nat ffect of the iffect of fla iffect of cxp	o of Application hod of application didity cension	$r_{o} = 0.05$ $m_{o} = jet$ $s_{o} = 300$ $e_{o} = 7$	5 gal/ft ² /min r m ₁ = sprav. -450 dynas/cm ² $e_1 = 14$.	1 = 0.075 gal/ft ² / 3 ₁ = 600-850 dyna	∕in r ₂ = 0.1 ga a∕cn ² .	l/ft ² /min.			A Shap Shap Shap Shap Shap
Levels of si	gnificenco	Het 1 per cent	34 5 per cer	nt.		;		• ,		•
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APPENDIX 2

Estimation of evaporation losses from foam spray

The loss of water by evaporation as the foam passes through the flames will depend on the foam expansion and particle size and the time taken for the foam to pass through the flame.

Consider unit mass of fcam comprising N spherical flakes of initial radius r_0 . The rate of loss of weight is given by

.(1)

 $\frac{dM}{dt} = \frac{hNSH}{L} + \frac{INS}{L}$ $\frac{dK}{L} = \frac{hNSH}{L} + \frac{INS}{L}$ $\frac{h}{L} = \frac{hNSH}{L} + \frac{INS}{L}$

Data on heat transfer coefficients between spheres and air have been correlated in terms of the Reynolds No. (Re) and the Nasselt No. (Nu) (4). To simplify the integration of equation (1) the following correlation will be used -

Nu = 0.63 Re 0.5

where

This gives a good correlation over the range of Re from (20-2,000) which is the range concerned in this problem.

Substituting in Equation (1) $\frac{dM}{dt} = \frac{0.63K}{2r} \left(\frac{2vr\rho}{\mu} \right)$ = thermal conductivity of the hot gases where = density μ = density μ = viscosity \mathbf{v} = relative velocity between the foam flakes and the flame.

It is reasonable to assume that the flakes will be travelling at their terminal velocity through the flames. The terminal velocity 0.5(Re > 500)V co $f_{F}^{0.7}$ (Re < 600) only $V \propto A^{0.5} \times 0.5(Re > 500)$ The range of flake sizes up to about 1 cn diameter would be in the first

The range of flake sizes up to about 1 cn diameter would be in the first range of Re. In determining the residence time of the flakes in the flames calculated values of settling velocity will be used, but since in equation (2) \vee appears only as $\vee^{0.5}$ the following approximation will be used -

 $V = A \rho_F^{0.7} \mu$ where $A = 9.9 \times 10^3$

Substituting in equation (2)

ρ_F^{0.35}3γ²0 + γ₀³ρ_FL $-\frac{dM}{dt} = \frac{0!}{2}$ + 0.945 KOp Now r = ro M 3

Integrating gives

$$M^{\prime 3} = I - \frac{t}{3r_{0}\rho_{F}L} \left[3I + 0.945 K \theta \rho_{F}^{0.35} \left(\frac{2A\rho}{\mu} \right)^{0.5} \right] \dots (3)$$

If it is assumed that the flakes always traverse the same distance through the flames the residence bind in the flames (t) will depend on the terminal velocity with respect to the petrol surface (\vec{V}) which depends on the flame velocity (VF) and also on r_0 and $f_{\rm F}$.

V = V - V_F

Since r will vary as the flakes pass through the flames the terminal velocity vall be continually varying. However, a 50% loss in weight of the foam flake would only result in a 16% reduction in r and thus a 16% decrease in velocity and a proportional incrase in residence time. A reasonable approximation can therefore be obtained by taking the terminal velocity as that appropriate to the initial flake size.

The evaporation losses for foam sprays of expansion 7 and 14 will be computed assuming the following values for the constant in Equation (3).

> L = 620 cal/gm. θ = 1000°C $K_{1.3 \times 10^{-4}}$ c.g.s. units $f_{1000°C} = 2.8 \times 10^{-4}$ c.g.s. units $\mu_{1000°C} = 4.8 \times 10^{-4}$ c.g.s. units I = 3.6 cal/cm²/soc. (assumes an emissivity of 1). $A = 9.9 \times 10^{3}$ $C = \frac{1 \text{ length of path of f lakes thro' f lame}}{V - VF}$ secs

Assuming the flakes traverse 10 ft (305 cms) and $\ddot{V}_F = 305$ cm/sec.

then t =
$$\frac{3\gamma 5}{V - 305}$$
 scos

Substituting in Equation (3) gives

$$M_{3}^{*} = 1 - \frac{0.165}{(v - 305) v_{0} \rho_{F}} \left[10.8 + 13.2 \rho_{F}^{0.35} \right]$$

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Thus for foams of expansion 7 -

$$M = \left[1 - \frac{20.1}{(V-305)} r_0^2 \right]$$

and for foams of expansion 14

$$M = \left[1 - \frac{37.15}{(V-305)} r_0^2 \right]$$

Values of M have been computed from these two equations for values of ro up to 1 cm and are shown graphically in Fig. 6



IGI RESULTS OF "FOG-FOAM" TESTS ON 1250 FT² FIRE (rates of application calculated from nozzle delivery)





FIG. 3.

DS84072/1/2356010/57CL





FIG 5 MOST LIKELY RESULTS OF TEST



